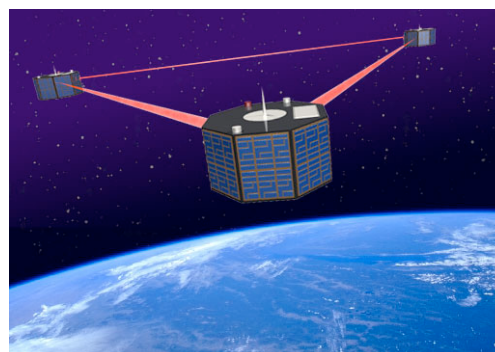
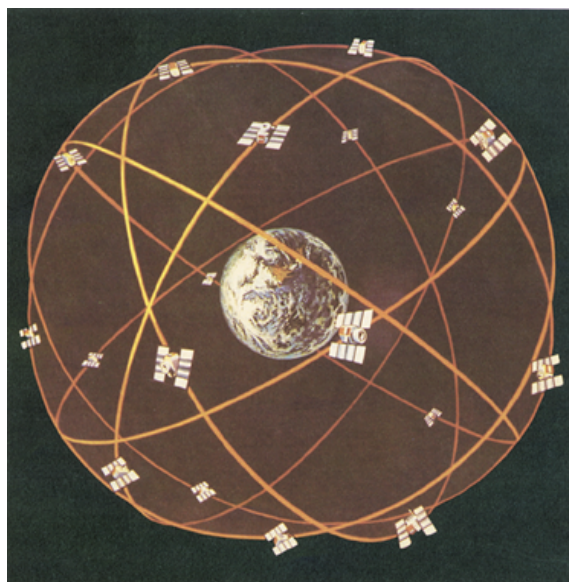
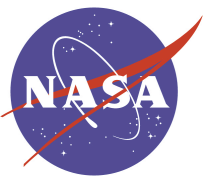


The Global Differential GPS System

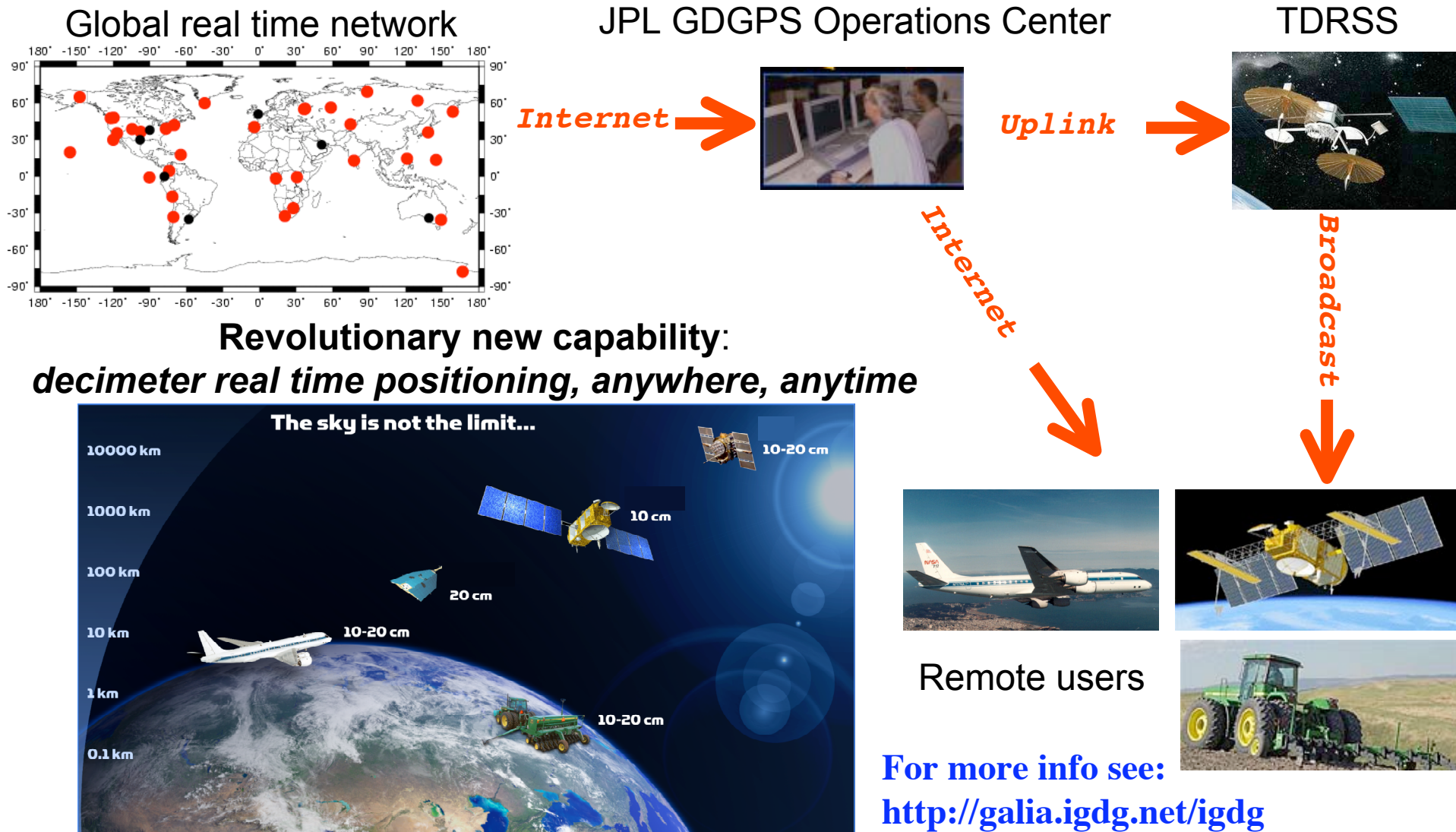
Yoaz Bar-Sever/JPL

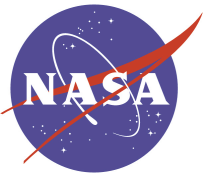
Yoaz.Bar-Sever@jpl.nasa.gov





JPL's Global Differential GPS System





Specific Benefits



Enable precision Interferometric SAR without crosslink metrology

- 0.1 nsec synchronization across arbitrarily large formations
- Bistatic SAR monitoring of currents and river flow
- Real time monitoring and assessment of natural hazards
- Reconfigurable and repairable radar apertures



Enable precision orbit control

- 10 cm repeat path accuracy from air and space
- Provides GPS integrity for autonomously controlled satellites

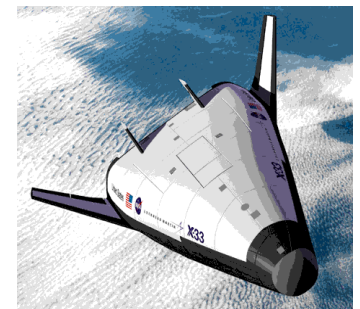


Improved Earth system forecasting capabilities

- Improved weather forecasting and tactical oceanography by lowering latency of Geophysical Data Records (GDRs) from altimetric satellites and GPS occultation constellations

Next generation shuttle/RLV/Space Plane

- cm-level autonomous proximity operations
- Safe autonomous landing anywhere in the world
- Proposed JPL/MSFC TCMIG navigation payload is based on GDGPS



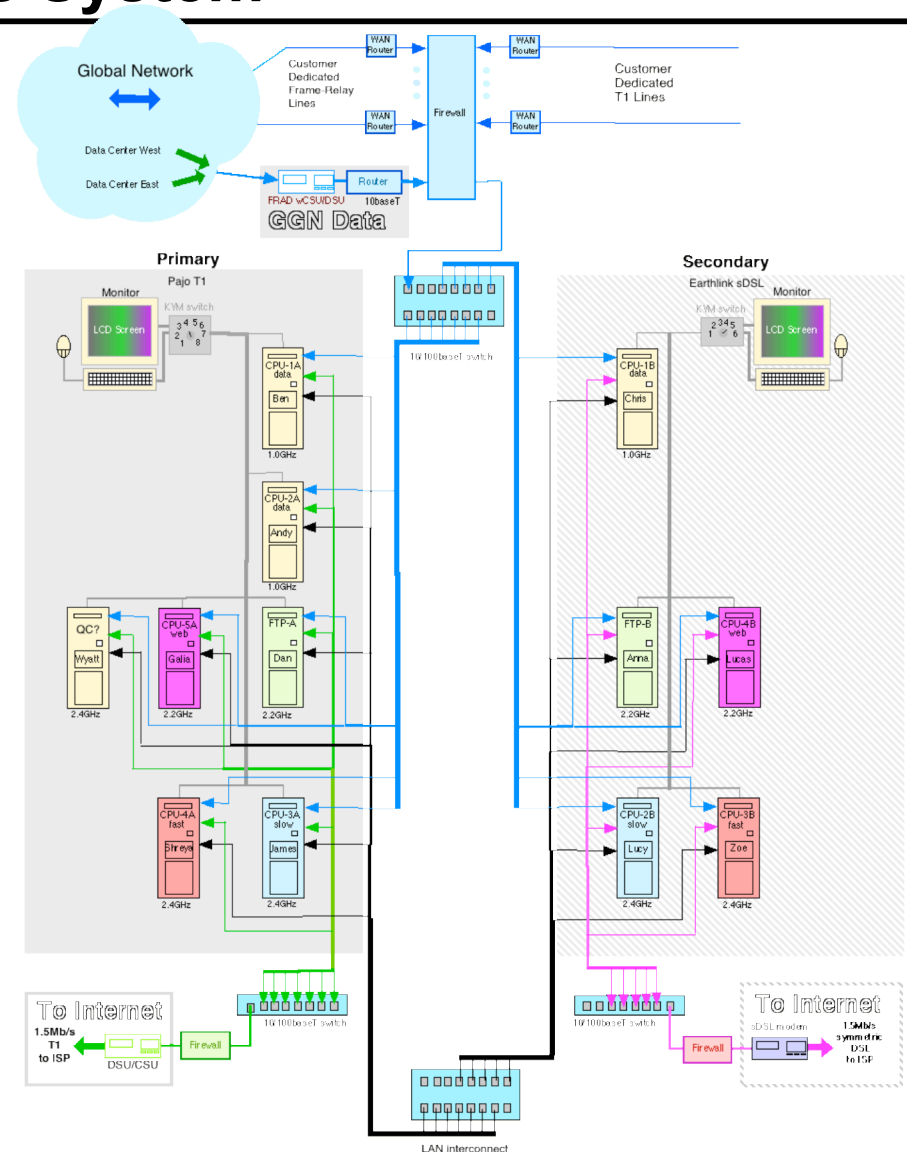
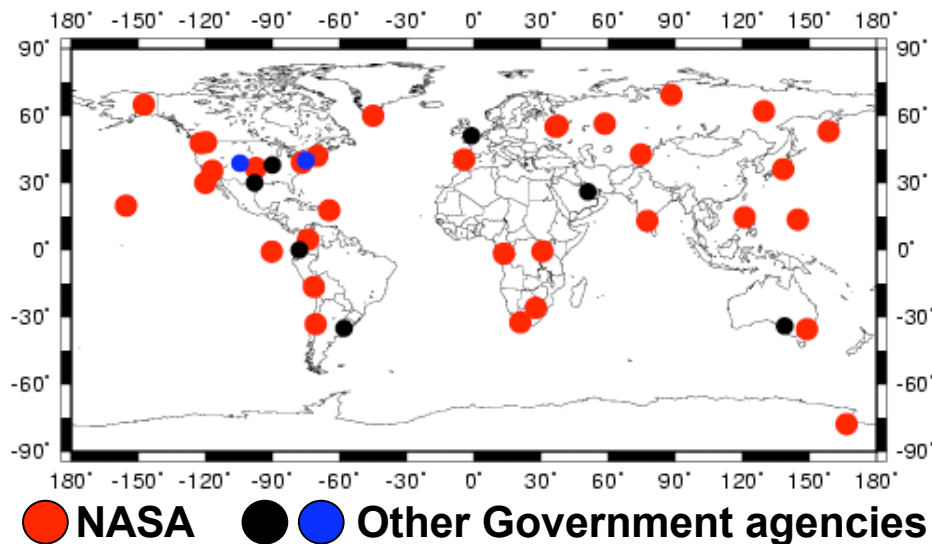
Many civil and national security applications



Mature and Reliable Ground Operations System



- Reliability through redundancy: **No single points of failure**
- Automatic fault detection and data rerouting
- Continuous Web monitoring in the public domain
- **99.99% reliability since 2000**



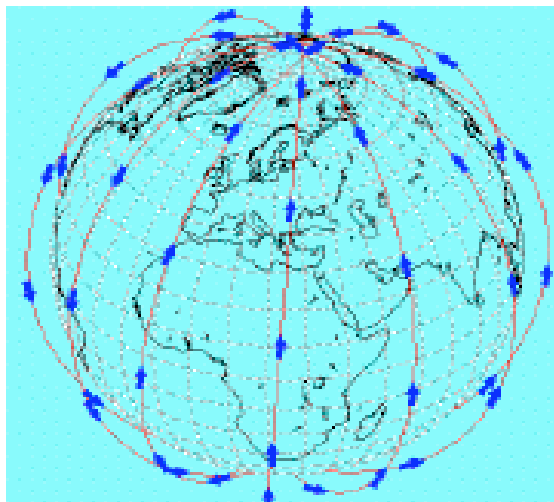
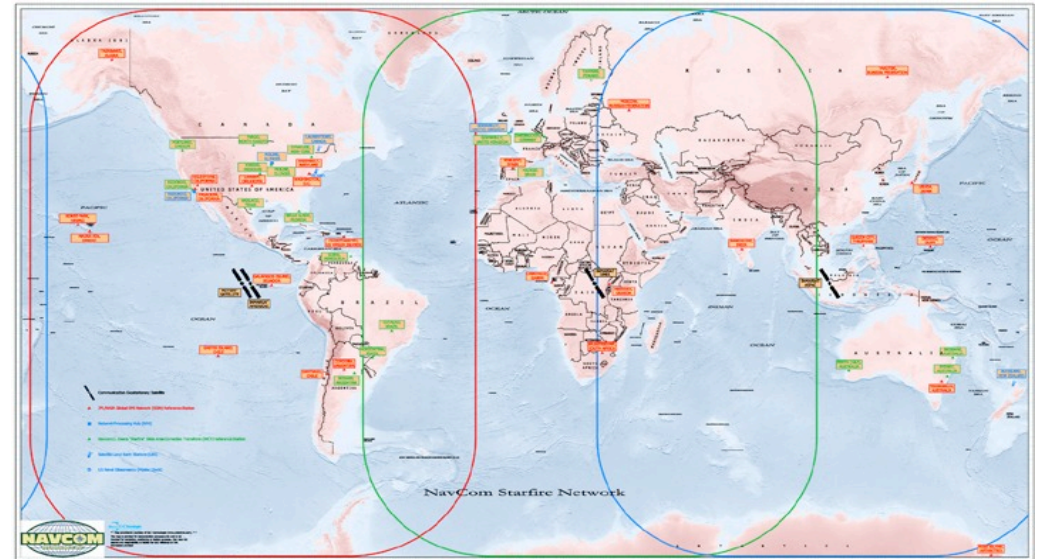


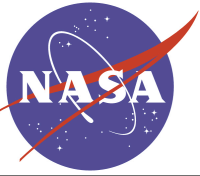
Communication Relay Systems



True global reach:

- The open internet
- 3 Inmarsat satellites with global coverage up to latitude $\pm 75^\circ$ (operated by Navcom)
- Iridium telephone modems provide internet access globally (including the polar regions)
- Future TDRSS broadcast

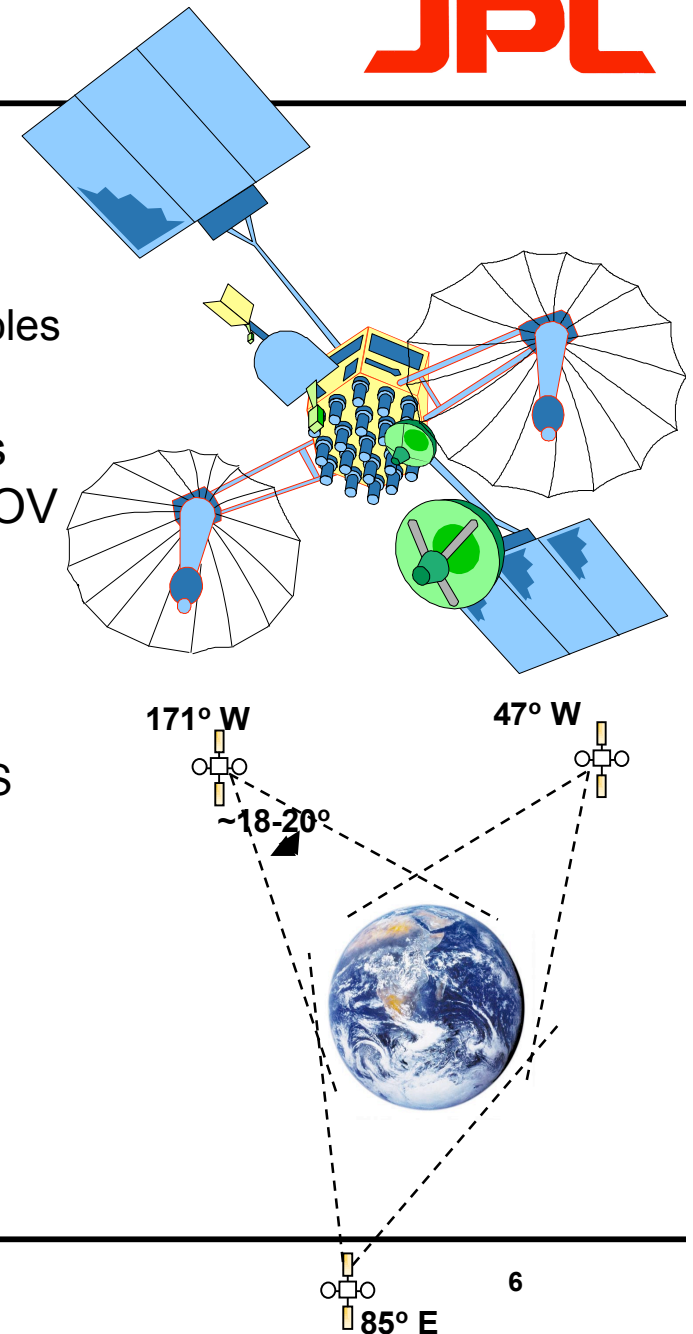




New TDRSS Service: GDGPS Broadcast



- New global beam designed
 - Element phasing controlled via upload commands
 - Fixed beam $\sim \pm 9 - 10^\circ$; covers altitudes to ~ 1000 km
 - Worst case link margin: 3 dB at 1000 km over the poles
20 dB over the equator
- 24 x 7 S-band MA broadcast from each TDRS provides continuous GDGPS correction data to all LEO s/c in FOV
- MA broadcast baseline
 - PN code modulated / BPSK data stream
 - Unique PN code per TDRS
 - 400 bps data rate including rate 1/2 FEC for 1 Hz GDGPS corrections
- Next Generation TDRSS will be more capable
 - Wider beam for better polar coverage
 - Stronger signal
- Initial ground tests in FY03; funded by Code M





Airborne Demonstrations

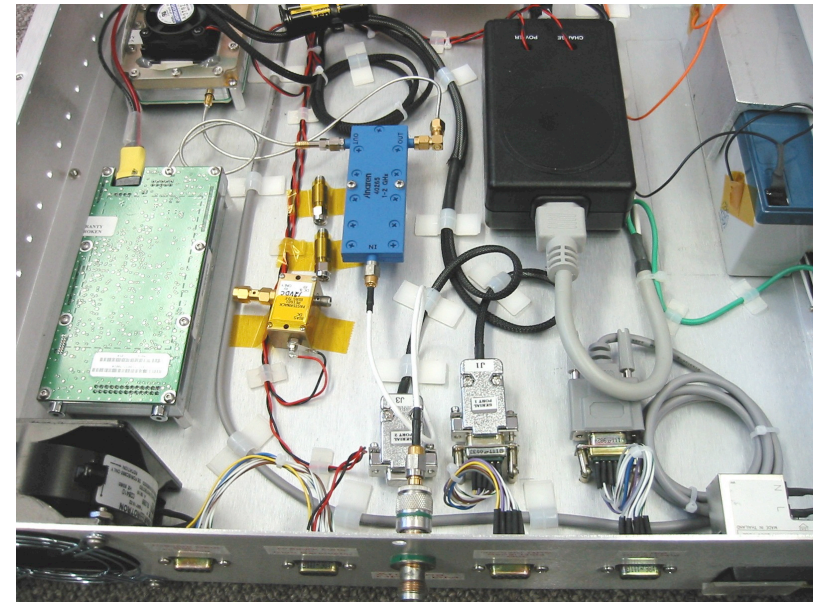


Extensive flight tests

- North America, February - September 2002: NASA DC-8 AirSAR
- Greenland, May 2002: NASA P-3 LIDAR
- Sweden, polar region, February 03: NASA DC-8

Performance validation by comparisons with:

- Post processing (precise orbits + smoothing)
- Independent local area differential techniques
- Laser ranging

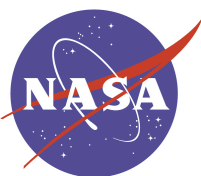


The NASA DC-8

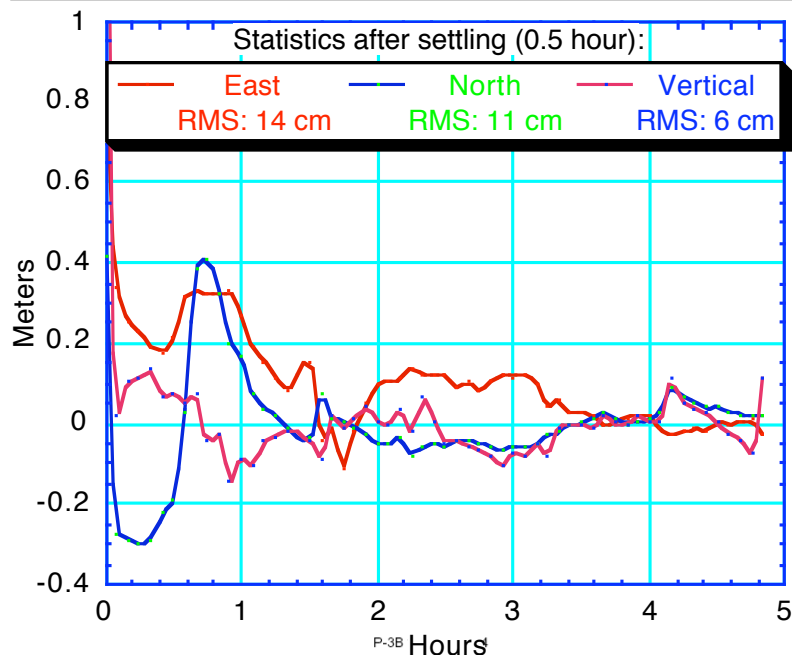


The NASA P-3

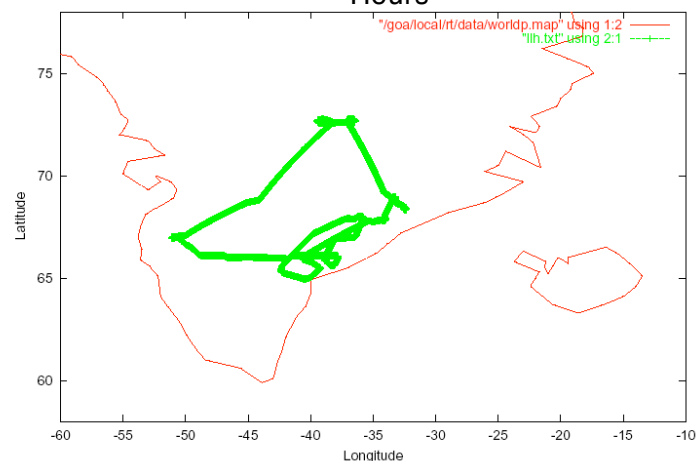
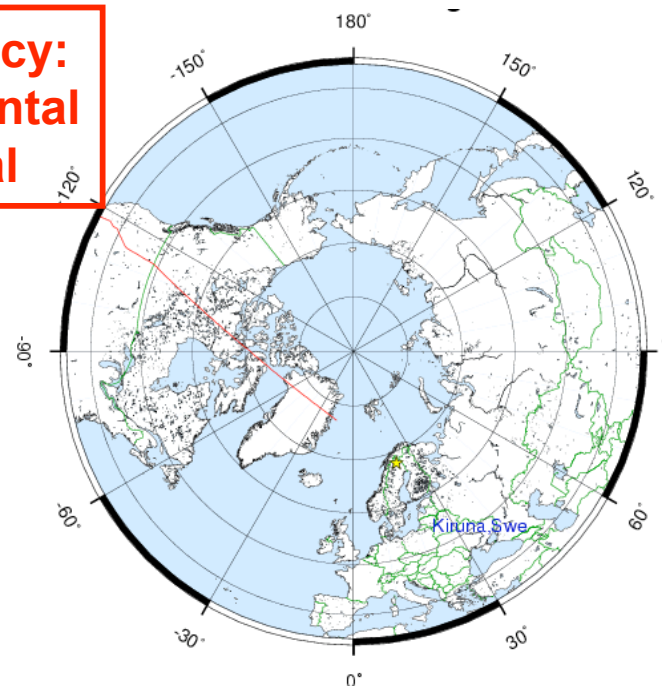




Airborne Demonstraions (cont.)

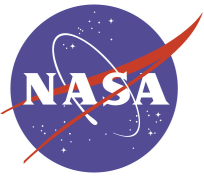


**Consistent Accuracy:
10 cm RMS Horizontal
20 cm RMS Vertical**



Difference between GDGPS and LADGPS (cm)

Date	East	North	Vert
31 May	17.2	10.1	13.0
02 Jun	4.5	4.6	5.9
04 Jun	8.0	6.1	12.6

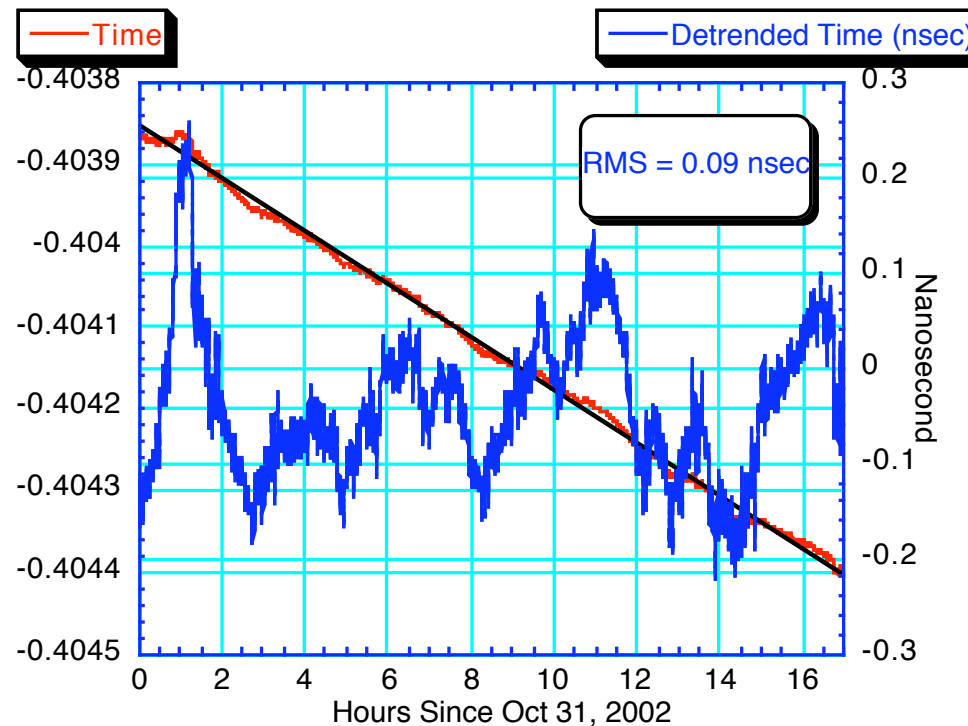


Time Transfer Capabilities

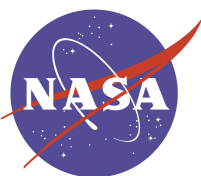


Real-Time time transfer accuracy is consistent with that of positioning:
nanosec level for dynamic platforms; 0.1 nanosec level for static platforms

Time Transfer from USNO to JPL Maser
Position held fixed



An order of magnitude improvement compared to unaided GPS



Orbit Determinations Demonstrations



Assess end-to-end system performance of precise real time orbit determination

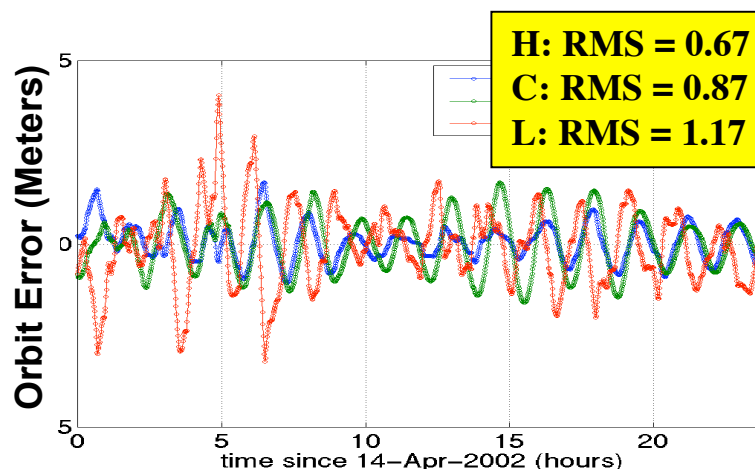
- In-space tests of software onboard SAC-C (but without GDGPS correction data)
- GDGPS performance “simulated” with *real data, real orbits, real time filter*

Altitude: 650 km

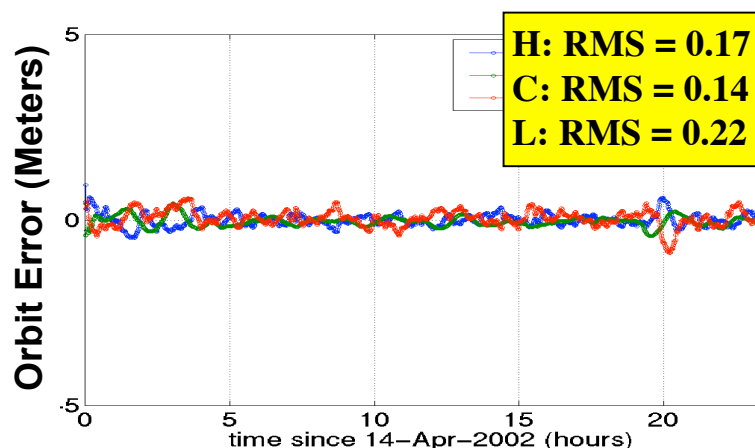


Results for better modeled satellites:

CHAMP: **20 cm** 3D RMS
Jason-1: **10 cm** 3D RMS
2.5 cm Radial
(science quality)



This is how well we currently perform onboard SAC-C (w/o GDGPS corrections)



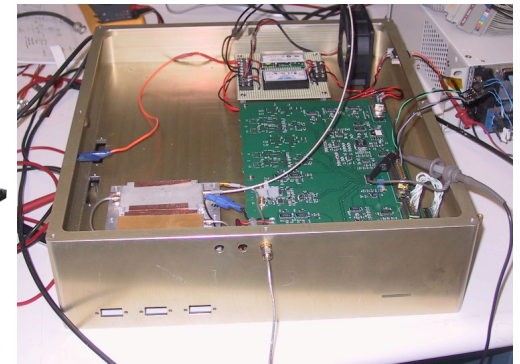
This is how well we expect to perform when the complete differential GPS receiver is deployed on a future SAC-C-like s/c



Spaceborne Global Differential GPS Receiver **JPL**

Hardware approach: flexible configuration that can be adopted to a variety of flight opportunities and risk postures

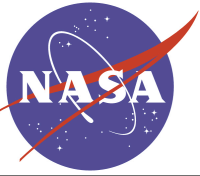
- A two-box configuration comprising of a BlackJack GPS receiver and a differential receiver/processor with S-band front-end and GDGPS software
 - Differential receiver is based on the Autonomous Formation Flyer baseband processor board
 - Fully reconfigurable FPGA; PowerPC 750
- A single box comprising of Blackjack GPS receiver, augmented with S-band front end and GDGPS software



Developed prototype spaceborne differential GPS receiver for precise real time orbit determination and time transfer

Development status:

- Completed design, fabrication, and assembly of a prototype differential receiver
- Receiver undergoing testing



The Need for Flight Validation



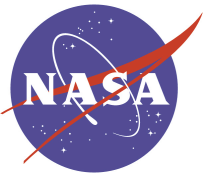
Payload cannot be adequately tested on the ground

- No hardware-in-the-loop simulators capable of decimeter level accuracy
- No hardware-in-the-loop simulators for combined GPS and GDGPS corrections

Orbital environment provides unique challenges

- Due to occultation of signal in troposphere and ionosphere
- Different Doppler signature
- Lower SNR
- TDRSS switchover

=> Flight validation is necessary to retire risk for first operational use




Flight Opportunities

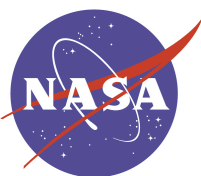


Goal: Identify missions of opportunity to host the GDGPS receiver

	<u>NASA</u>	<u>DoD/OGA</u>	<u>Foreign Collaboration</u>
2004			
2005		MLV-5	
2006	NPP		EQUARS
2007	OSTM*, OCO	LDCM	SAOCOM
2008	GPM, AQUARIUS, Ocean Vector Winds		
2009	Restless Planet*		

 * Mission enhancing

In bold: strong prospects; positive initial contact

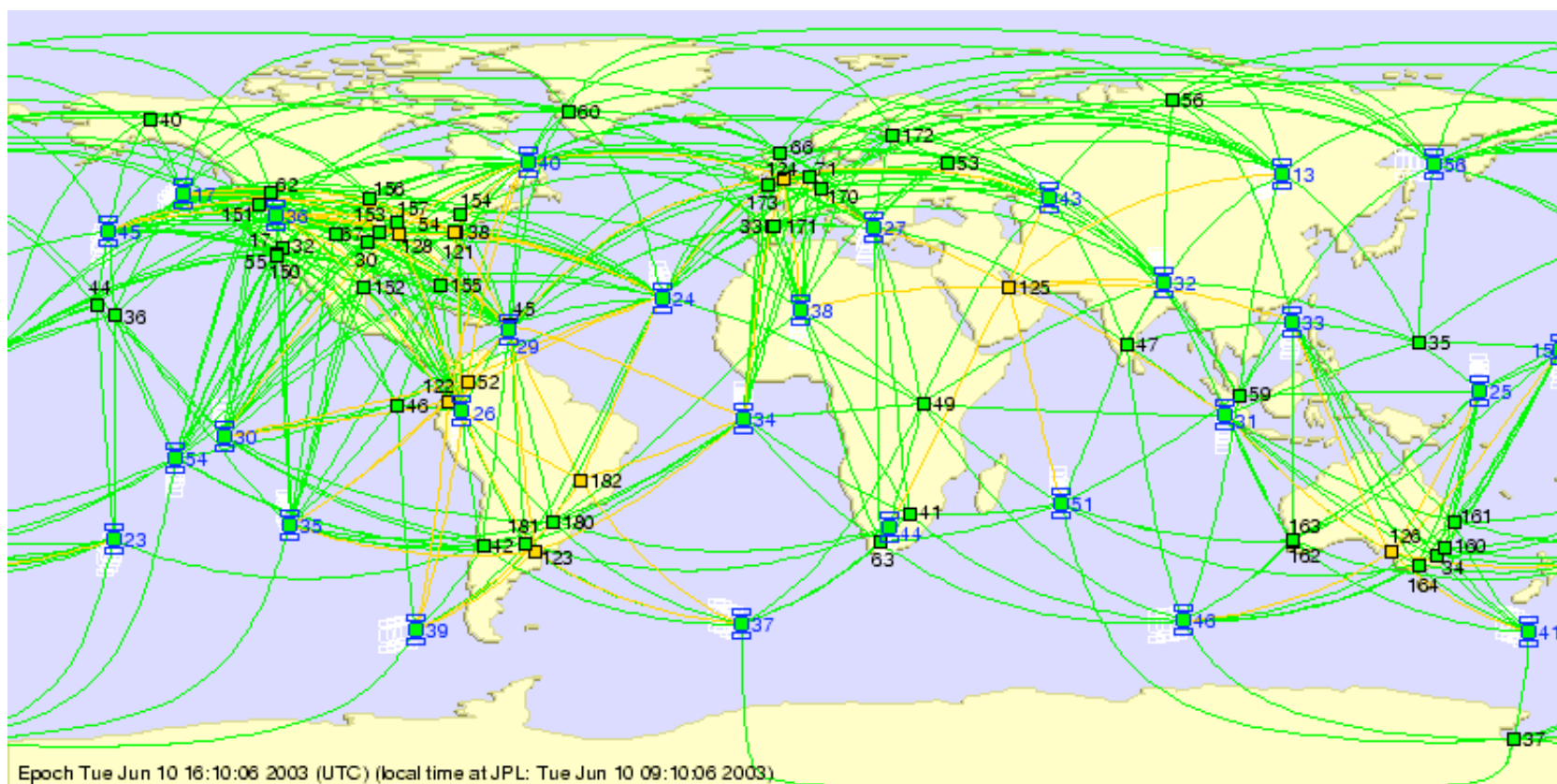


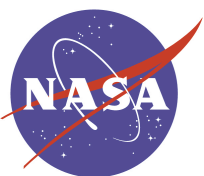
GPS Integrity Monitoring



GDGPS is the only redundant real time network observing all GPS satellites all the time

State-space approach enables separate monitoring of clocks and orbit states



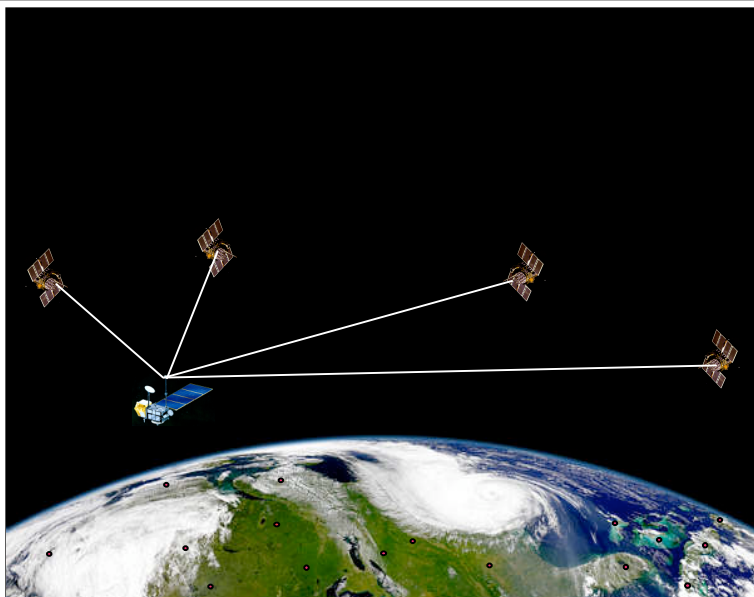


Revolutionary New Capabilities

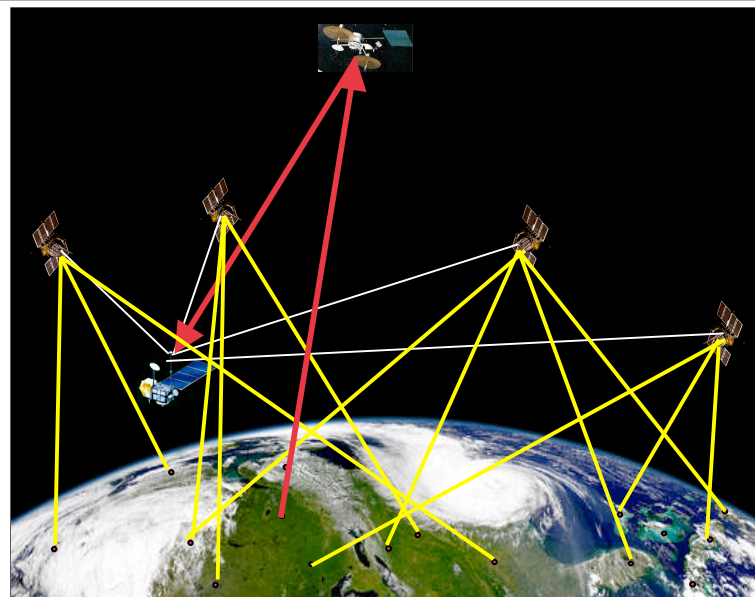


The NASA Global Differential GPS (GDGPS) system provides spacecraft with revolutionary new capabilities for real-time orbit determination and time-transfer

	State of the Art (unaugmented GPS)	GDGPS	The Next Step: GDGPS+IMU+Optical
Real-time orbit determination	Few meters	0.1 - 0.3 m	0.01 m
Real-time time-transfer	~10 nsec	0.1 nsec	< 0.1 nsec
Integrity (GPS malfunction flags)	Not available	Included	Included

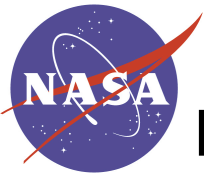


June 23, 2003



ESTO Conference

15



Roadmap to Autonomy and Reduced Latency

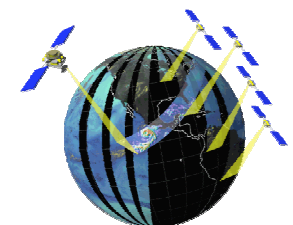
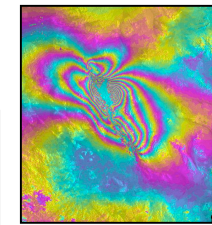
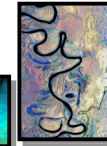
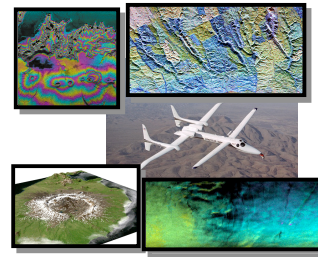
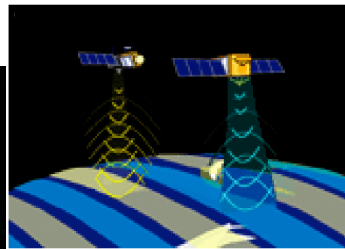
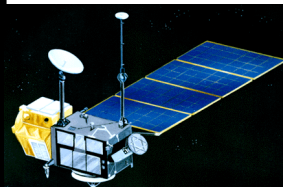


Goals

Reduce latency of science products

Real-time Earth monitoring; Autonomous spacecraft operations; Decimeter accuracy of real-time science products

Natural hazard forecasting and monitoring; constellations for weather and remote sensing; Centimeter accuracy of real-time science products



mission concept

Near real time processing on the ground with real-time GPS orbits

Onboard real-time position knowledge and control; Onboard science processing; Repeat pass interferometry

Coordinated measurements and constellation sensor autonomy; real-time precision formation flying; distributed data processing; L-band InSAR

New system architecture & technologies

Ground processing with NASA's Global Differential GPS System

Global Differential GPS receiver and software for onboard positioning and timing

TDRSS GDGPS service

GDGPS/accelerometer/star-tracker integrated receiver; Constellation Communications and Navigation Transceiver

Reconfigurable radar; Radar arrays

Increasing Science Capability and Time

